

Size Effect in Diagonal Shear Failure: Influence of Aggregate Size and Stirrups



by Zdeněk P. Bažant and Hsu-Huei Sun

Fracture mechanics of brittle failures of concrete structures due to the propagation of a cracking zone indicates that the nominal stress at failure should not be constant but should decrease as the structure size increases. This size effect may be described by a simple size-effect law recently derived by dimensional analysis and similitude arguments. A formula for diagonal shear failure based on this size-effect law was published, verified, and calibrated by a large set of data from the literature. The present paper improves this previous formula in two respects: (1) the effect of maximum aggregate size, distinct from the effect of the relative beam size, is incorporated and (2) the formula is extended to cover the effect of stirrups on the shear capacity of concrete. The new formula is verified and calibrated according to a larger data set than before, consisting of several hundred test results compiled from the literature. It is shown that the new formula achieves an appreciable reduction in the coefficient of variation of the deviations of the measured data from the prediction formula.

Keywords: aggregate size; beams (supports); cracking (fracturing); crack propagation; failure; loads (forces); optimization; reinforced concrete; shear properties; statistical analysis; stirrups; tests.

The diagonal shear failure of reinforced concrete beams is a classical yet formidable problem that has not been resolved to complete satisfaction despite several decades of study.¹⁻³⁷ Although the design formulas for diagonal shear have been gradually improved, they still exhibit large errors that are in fact much greater than the random errors in the measurements of strength or of fracture energy. So there is clearly room for further improvement.

Until recently, the formulas proposed for diagonal shear failure were either purely empirical or based on plastic limit analysis. A new viewpoint was introduced in 1981 by Reinhardt,^{38,39} who suggested that the design formula be based on linear elastic fracture mechanics and substantiated his suggestion by certain test results. Subsequently, it was established^{40,41} that the size effect implied by linear elastic fracture mechanics is too strong in the case of concrete, and that brittle failures of concrete structures are better described by nonlinear fracture mechanics, which is based on the existence of a large cracking zone at the fracture front and yields a considerably weaker size effect.

At the same time, a simple approximate size-effect law for brittle failures due to fractures that are blunted by a zone of distributed cracking was derived⁴¹⁻⁴⁴ and shown to be approximately applicable to various brittle failures of concrete structures, including the present problem of diagonal shear failure in both non-prestressed⁴⁵ and prestressed beams,⁴⁶ the punching shear failure of slabs,⁴⁷ the ring and beam failures of unreinforced pipes,⁴⁸ and torsional failures of longitudinally reinforced beams.^{49,50} A formula for diagonal shear failure of beams without stirrups, incorporating the size-effect law, was developed⁴⁵ and shown to offer significantly reduced errors in comparison to the existing formulas of ACI⁵¹ or CEB-FIP,⁵² which were developed on the basis of crack initiation rather than failure.

The purpose of the present study is to: (1) improve the previously proposed formula based on the size-effect law by incorporating, in addition to the effect of relative beam size, the effect of aggregate size; (2) recalibrating the formula with a larger data base, encompassing essentially all the adequately documented test results that can be found in the literature; and (3) extending the formula to beams with stirrups.

PREVIOUS FORMULA INCORPORATING THE SIZE-EFFECT LAW

In a previous paper,⁴⁵ Bažant and Kim derived and justified by existing test data the following formulas for the diagonal shear failure of beams with longitudinal reinforcement but without stirrups [Fig. 1(a)]

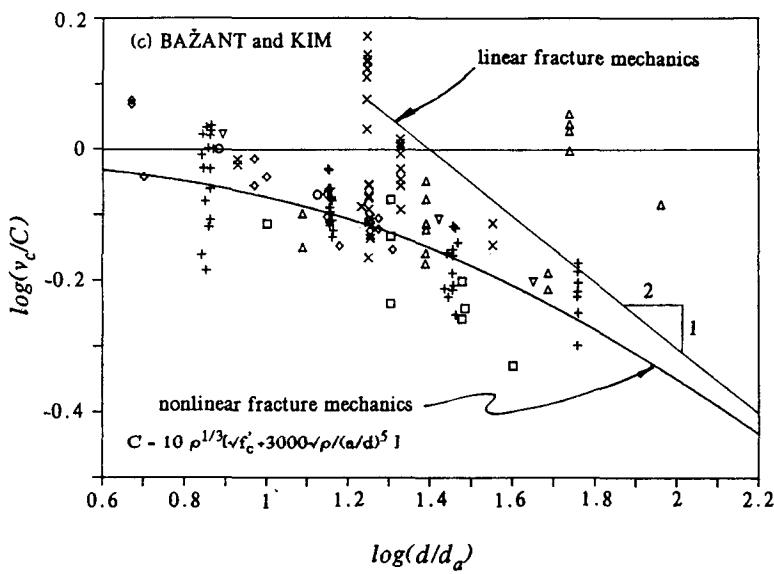
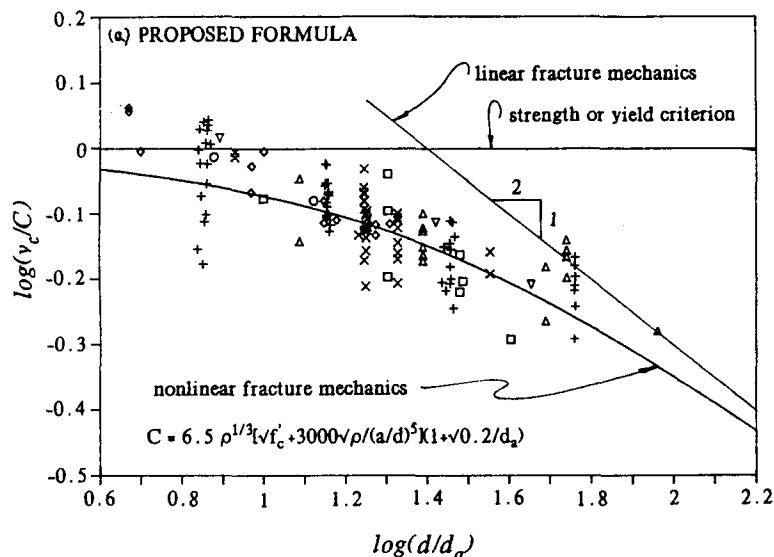
$$\nu_c = \nu_c^0 \left(1 + \frac{d}{\lambda_0 d_a} \right)^{-1/2} \quad (1)$$

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$$v_c^0 = k_1 \rho^{1/3} \left(\sqrt{f'_c} + 3000 \sqrt{\frac{\rho}{(a/d)^5}} \right) \quad (2)$$



where

v_c = nominal shear strength at failure

V_c = shear force at failure

b = beam width

d = beam depth measured to the axis of longitudinal reinforcement

d_a = maximum aggregate size

a = shear span (equal to the distance of the concentrated load from the support when a simply supported beam is loaded symmetrically by two concentrated loads)

ρ = steel ratio for longitudinal reinforcement

f'_c = standard cylindrical compressive strength in psi (6895Pa)

k_1, λ_0 = empirical constants ($k_1 = 10$)

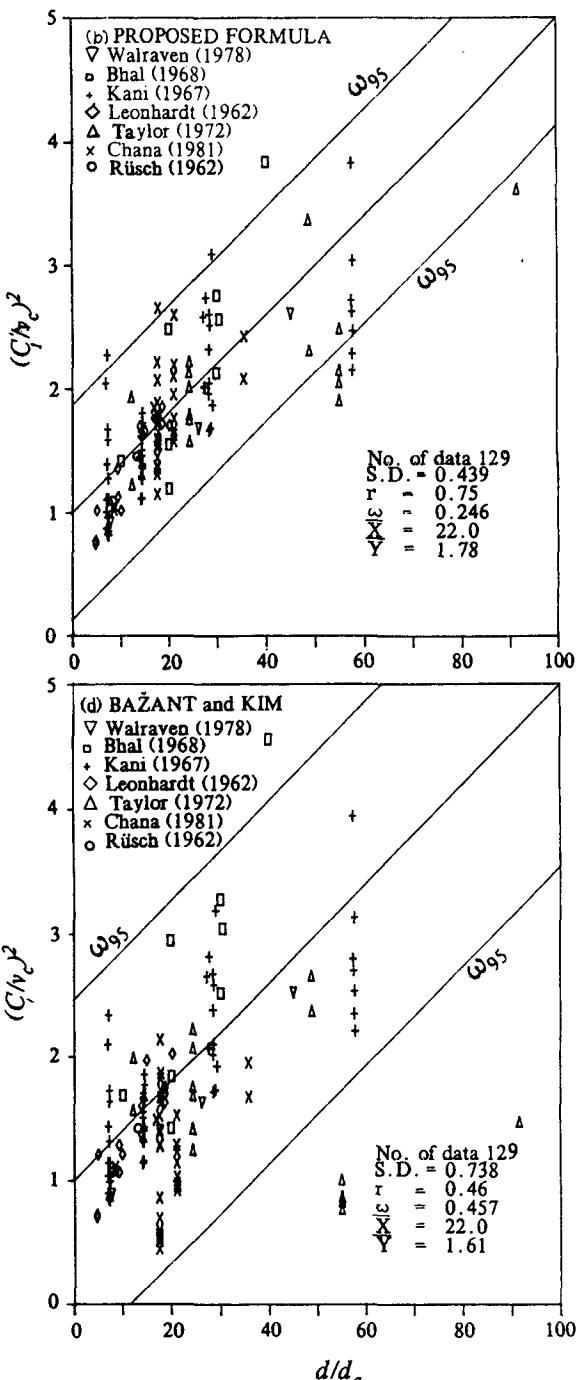


Fig. 1—Effect of beam size in beams without stirrups

Eq. (2), which is quite similar to the formula currently used by ACI, has been derived by analyzing the arch action and the composite beam action, and summing their contributions.⁴

The size effect is due to the release of strain energy from the beam into the cracking zone as the cracking zone extends; the larger the structure, the greater is the energy release. Eq. (1) expresses Bažant's size-effect law,⁴² which was derived by dimensional analysis and similitude arguments from two simplifying assumptions, namely that the energy loss due to cracking is a function of both the fracture length and of the area of the cracking zone assumed to have a constant width at its front, proportional to the maximum aggregate size. Although this size-effect law is approximate, it was shown to capture the basic trend correctly when geometrically similar specimens of different sizes, failing in a brittle manner, are considered.

For very small structure sizes d , such that $d/\lambda_0 d_a \ll 1$, Eq. (1) approaches the limiting case of plastic limit analysis, which is characterized by absence of the size effect. For very large structure sizes d , such that $d/\lambda_0 d_a \gg 1$, Eq. (1) approaches the limiting case of linear elastic fracture mechanics. For normal beam sizes, Eq. (1) describes a smooth transition between these two limit cases.

Applicability of the size-effect law in Eq. (1) to structures that contain no notches is further contingent upon the assumption that the failure surfaces (cracks) in beams of various sizes are geometrically similar, and that the failure occurs when the front of the final crack reaches approximately the same relative position within the beam regardless of the size. These assumptions appear to agree with experimentally observed behavior.

It should be mentioned that the most general size-effect law for geometrically similar structures or specimens may be described by an asymptotic series expansion presented in Reference 43. The approximate size-effect law in Eq. (1) is then found to represent the first two terms of this asymptotic series expansion. It has been also shown^{45,53} that an improved size-effect law for blunt fracture can be obtained as $v_c = v_c^0 (1 + \xi)^{-1/r}$ with $\xi = d/\lambda_0 d_a$, in which r is an empirical constant. In the course of the present study, however, it appeared that the existing test data for concrete, due to their scatter as well as limited range, are insufficient for determining r . The optimum values of r for the existing diagonal shear data for beams ranged from 0.75 to 1; the fits were about equally good for this entire range. Therefore, the value $r = 1$ has been adopted.

Eq. (1) and (2) involve no size effect of beam width. This means that the cracking is assumed to occur roughly simultaneously through the thickness rather than propagate across the thickness.

Fracture mechanics is not the only theory that yields a size effect. A competing theory is the statistical Weibull theory. Although both theories can describe the same size-effect data if their size range is rather limited, they have very different extrapolations to very large sizes. While Weibull theory is certainly applicable

to brittle failures of structures that can be modeled as a chain of brittle suddenly failing elements, it is questionable to apply it to diagonal shear failure.⁴⁴

There is no aspect in this problem that could be regarded as a failure of a one-dimensional chain of elements. The release of strain energy into a localized extending fracture front, which is the basis of fracture mechanics, has nothing to do with Weibull's assumptions. It has also been pointed out⁴⁴ that if the material parameters of the extreme value strength distribution underlying Weibull theory are calibrated by typical test data for homogeneously stressed tensile specimens, and the same material parameters are then used for the cracking front zone, the statistical Weibull-type contribution to the size effect is found to be negligibly small, due to the fact that the volume of the frontal cracking zone is rather small compared to the beam size.

In very large beams, significant size effect may also be caused by diffusion phenomena, such as conduction of hydration heat and diffusion of pore water. Larger beams heat up more, and they also lose moisture more slowly than small beams. For lack of information, the diffusion size effects must be ignored in the analysis of existing test data.

Eq. (1) and (2) were calibrated⁴⁵ by a set of 296 data points extracted from the literature. Unfortunately, most of these data were not obtained with the size effect in mind, and therefore, the conclusions for the size effect cannot be definite. Nevertheless, plotting the existing data in the proper variables, the existence of the size effect was clearly demonstrated and it was shown that the size effect did not disagree with the theoretical size-effect law [Eq. (1)]. The scatter of the data was large, since it was necessary to include in the comparison data obtained in different laboratories, with different concretes and test beams that were not geometrically similar. This produces a scatter of the data that is large and obfuscates the precise trend with regard to the beam size.

As for comparisons with the existing formulas of ACI⁵¹ and CEB-FIP,⁵² the improvement achieved by Eq. (1) and (2) was significant; the standard deviation of the errors in v_u compared to the formula was found to be 128 psi for the ACI formula, 148 psi for the CEB-FIP formula, and 44.6 psi for Eq. (1) and (2). On the other hand, the improvement over another previous excellent formula due to Zsutty^{18,21} was negligible when the entire data set was considered. However, Zsutty's formula gives no size effect and the presence of the size effect is clearly verified by those scant data in which beams of significantly different sizes were included in the test series (see Fig. 5 of Reference 45).

PROPOSED GENERALIZED FORMULA

To cover the effect of maximum aggregate size d_a , Bažant⁵³ derived by fracture mechanics considerations the following generalization of Eq. (1)

$$v_c = v_c^0 \left(1 + \sqrt{c_0/d_a} \right) \sqrt{1 + \frac{d}{\lambda_0 d_a}} \quad (3)$$

where c_0 = empirical material constant having the dimension of length. By optimizing the fits of the test data described in the sequel, the optimum values of the empirical constants were found to be $c_0 = 0.2$ in., $\lambda_0 = 25$ and $k_1 = 6.5$. Note that for the aggregate size $d_a = 0.69$ in. (17.5 mm), Eq. (2) and (3) are equivalent to the original Eq. (1) and (2). In Reference 53 it was also shown that the factor $1 + (c_0/d_a)^{1/2}$, which gives the effect of aggregate size, agrees reasonably well with the test results of Taylor²⁵ and of Chana,³⁰ which include a wide range of maximum aggregate sizes.

With regard to the identification of material parameter values, Eq. (3) has the advantage that it can be algebraically transformed to a linear regression plot. This plot is described by the equation $Y = AX + C$ in which

$$X = \frac{d}{d_a} \quad Y = \left[\frac{1}{v_c} \left(1 + \sqrt{\frac{0.2}{d_a}} \right)^2 \right]^{1/2} \quad (4)$$

with $C = 1/v_0^2$ and $A = C/\lambda_0$. By linear regression of the test data plotted as Y versus X , one finds the slope A and the Y -intercept C of the regression line, and the material parameters are then obtained as $v_0 = 1/C$ and $\lambda_0 = C/A$. This regression plot reveals the effect of the relative size d/d_a .

Another algebraic rearrangement of Eq. (3) yields a linear regression plot $Y' = A'X' + C'$ in which

$$X' = \frac{1}{\sqrt{d_a}} \quad Y' = v_c \left(1 + \frac{d}{\lambda_0 d_a} \right)^{-1/2} \quad (5)$$

with $C' = v_0^0$ and $A' = v_{c_0} \sqrt{c_0}$. After determining the slope and the Y' -intercept of this plot, the material pa-

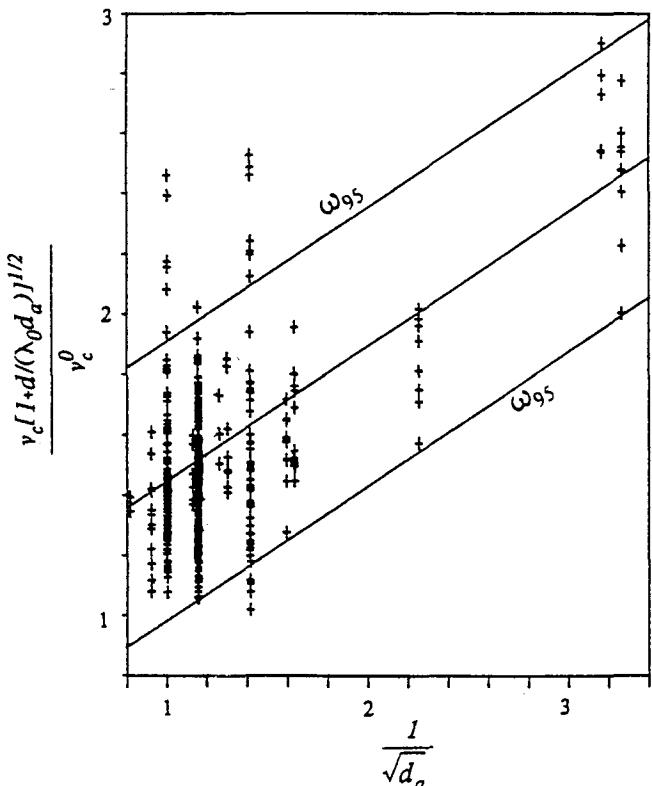


Fig. 2—Effect of aggregate size in beams without stirrups

rameters may be identified as $v_c^0 = C'$ and $c_0 = (A'/v_0^0)^2$. This type of regression reveals the effect of the maximum aggregate size.

To verify the present formulation, one would ideally need test data for geometrically similar beams of very different sizes. Unfortunately, no such test data exist in the literature. Most of the published test results pertain to beams of the smallest possible size that can be made with a given aggregate. Only a few test series, those indicated in Fig. 1(b), give at least some information on the size effect, although the size range of these data is relatively limited. These data, obtained by Walraven,²⁸ Bhal,¹⁹ Kani,¹⁶ Leonhardt and Walther,⁸ Rüsch et al.,¹⁰ Taylor,²⁵ and Chana,³⁰ are plotted in Fig. 1(a) through (d).

The large scatter of these data is due principally to the fact that the tests were made in different laboratories, on different concretes, and on geometrically dissimilar beams. Consequently, the error involves not only the error of Eq. (3) but also the error of Eq. (2), which enters Eq. (3). Nevertheless, the cost of producing a sufficiently large data set in one laboratory, on the same concrete, and for geometrically similar beams, would be high. First, one should exploit the information already available, even though it was not obtained for the purpose of studying the size effect.

From Fig. 1(a) through (c), it is clear that a significant size effect exists, since otherwise the trend of the data in these figures would have to be horizontal. Furthermore, it is clear that the size effect according to linear fracture mechanics, represented by the straight line of slope $-1/2$ in Fig. 1(a) and (c), would be too strong. The size effect is intermediate between the strength or yield criterion (limit analysis) and linear elastic fracture mechanics. This corroborates the acceptability of the assumption of a blunted fracture front from which Eq. (1) is derived. The size-effect plot of the new formula in Fig. 1(a) [with the corresponding linear regression according to Eq. (4) and (5) and Fig. 1(b)] is compared to the size-effect plot of the previous Bažant-Kim formula [Fig. 1(c) and Eq. (1)], with corresponding linear regression [Fig. 1(d)].

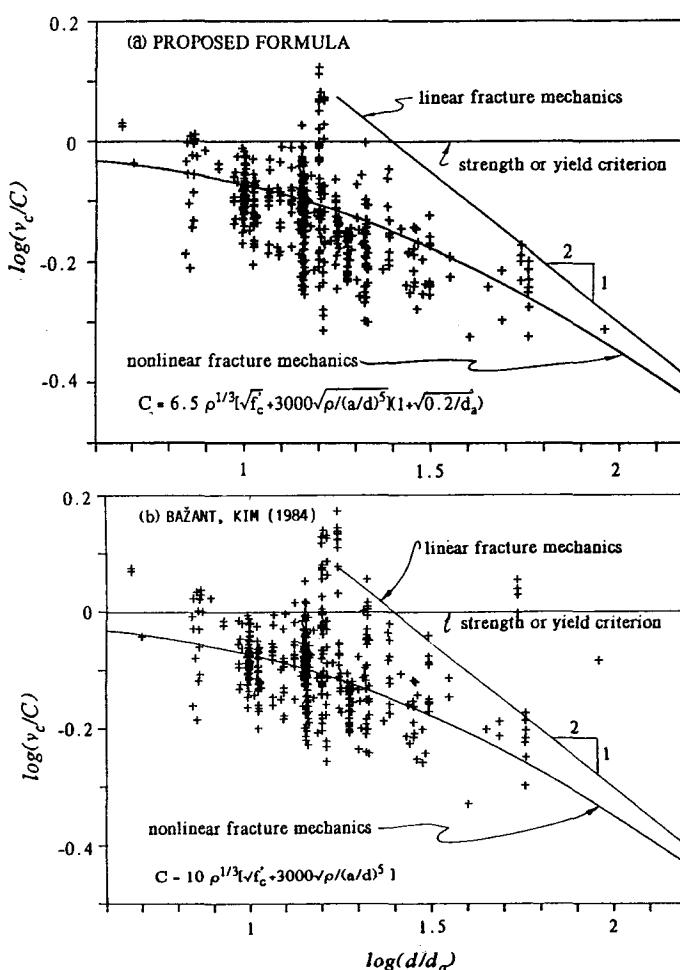
It is seen that the new formula, which includes the effect of aggregate size, significantly reduces the scatter. This is confirmed by the values of the coefficient of variation $\omega = S.D./\bar{Y}$ for the vertical deviations of the data from the regression line, listed in Fig. 1, or the regression correlation coefficient r . The notations are: S.D. = standard deviation of the vertical differences from the regression line; \bar{X}, \bar{Y} = coordinates of the centroid of the data; the lines of ω_{95} represent the 95 percent confidence limits; and, in Fig. 1, $C_1 = v_0 [1 + (0.2/d_a)^{1/2}]$. Compared to the previous Bažant-Kim formula, the present improved formula reduces the coefficient of variation of the errors from 0.457 to 0.246, and increases the correlation coefficient from 0.46 to 0.75 [Fig. 1(b) and (d)].

The linear regression plot of the same data, showing the effect of maximum aggregate size and based on Eq. (5), is presented in Fig. 2. Despite the very large scat-

ter, it is evident that, in addition to the effect of the relative beam size d/d_a , there is also an effect of the maximum aggregate size. If there were none, the trend of the data in Fig. 2 would have to be horizontal.

Fig. 3(a) and (b) show comparison of the present formulas [Eq. (2) and (3)] with 461 test results that have been extracted from the literature. In addition to the data used in Fig. 1, this large data set also includes the test results of Ahmad et al.,³⁷ Bhal,¹⁹ Bresler and Scordelis,¹¹ Chana,³⁰ Diaz de Cossio and Siess,⁶ Elzanaty et al.,³⁶ Gaston, Siess, and Newmark,⁴ Kani,^{15,16} Krefeld and Thurston,¹⁴ Leonhardt and Walther,⁸ Mathey and Watstein,¹² Mattock,²⁰ Mattock and Wang,³³ Moody et al.,⁵ Mphonde and Franz,³⁴ Placas and Regan,²⁴ Rajagopalan and Ferguson,¹⁷ Swamy and Bahia,³⁵ Swamy,³¹ H. Taylor,²⁵ Taylor and Brewer,^{7,13} van den Berg,⁹ and Walraven.²⁸

Since the compilation of these data has been a rather tedious process (requiring correspondence with some of the authors to obtain various missing information), the complete set of 461 data points is listed numerically in Table 1. This makes it possible to verify the present rules as well as to use these data for possible future improvements of the diagonal shear formula.

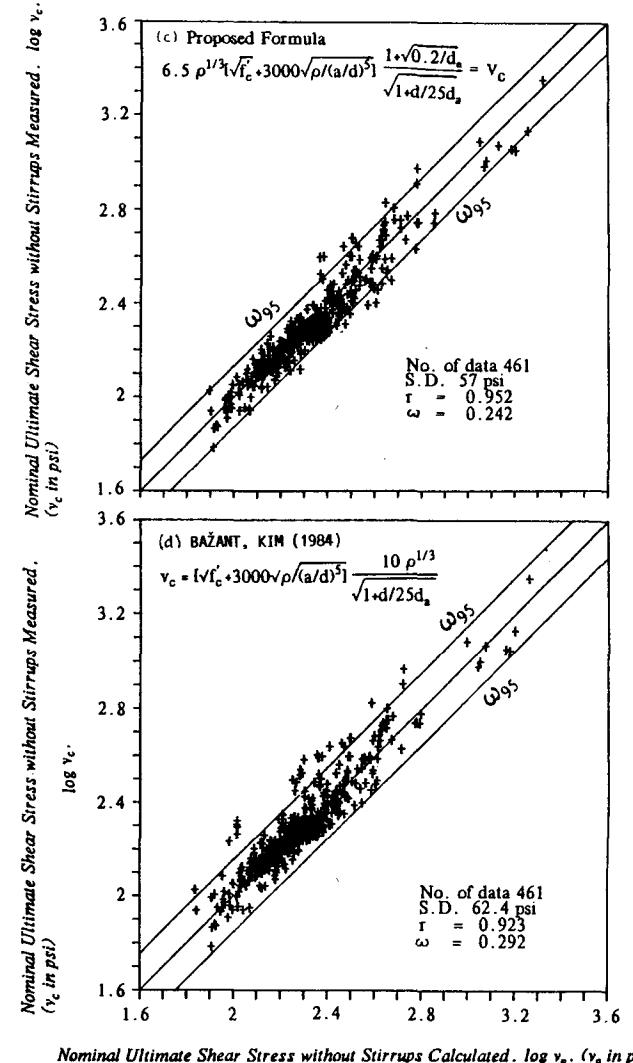


Since most of the data in the complete set of 461 data points in Fig. 3 do not cover the size effect, the effect of the relative beam size d/d_a is not as conspicuous as in Fig. 1, but from the size-effect plot in Fig. 3(a) it is still clear that a size effect exists. Fig. 3(c) shows a plot of the measured values versus the calculated values of the nominal shear stress at failure. If there were no scatter and the formula were perfect, this plot would be a straight line of Slope 1, passing through the origin. So the deviations from this straight line are a measure of the error.

For comparison, Fig. 3(b) and (d) show the same plots for the previous Bažant-Kim formula. We see that the present formula achieves a reduction of scatter, diminishing the coefficient of variation of the errors ω from 0.292 to 0.242, and increasing the correlation coefficient r from 0.923 to 0.952. Again, the lines marked ω_{95} represent the 95 percent confidence limits.

GENERALIZATION FOR THE EFFECT OF STIRRUPS

For a beam with both longitudinal reinforcement and stirrups, the nominal ultimate shear stress at failure may be expressed as



Nominal Ultimate Shear Stress without Stirrups Calculated. $\log v_c$. (v_c in psi)

Fig. 3—Plots of measured versus calculated values of nominal shear strength for 461 available data for beams without stirrups (1 psi = 6895 Pa)

Table 1—Test data from the literature

Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (in ²)	V ^c (lb)	V _c (lb)	Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (in ²)	V ^c (lb)	V _c (lb)	
1) Without stirrups, 3-point-bent																		
Moody, K. G. et al. (1954)																		
A1	31.5	7.	10.30	1.	4400	1.56	13000	13500		A0-3-3b	42.	6.	11.75	.375	3273	2.37	12750	14500
A2	31.5	7.	10.50	1.	4500	1.58	15000	15000		A0-3-3c	42.	6.	11.75	.375	4277	1.64	15000	15000
A3	31.5	7.	10.55	1.	4500	1.64	14000	17000		A0-7-3a	42.	6.	11.75	.375	5938	2.37	18500	18500
A4	31.5	7.	10.63	1.	4570	1.76	15000	16000		A0-7-3b	42.	6.	11.75	.375	6562	2.37	18600	18600
B1	31.5	7.	10.50	1.	3070	1.19	11450	12650		A0-11-3a	42.	6.	11.75	.375	11812	2.37	20150	20150
B2	31.5	7.	10.55	1.	3130	1.20	13500	13500		A0-11-3b	42.	6.	11.75	.375	11766	2.37	20100	20100
B3	31.5	7.	10.63	1.	2790	1.19	12000	12500		A0-15-3a	42.	6.	11.75	.375	12823	2.37	21000	21000
B4	31.5	7.	10.69	1.	2430	1.24	11900	12500		A0-15-3b	42.	6.	11.75	.375	14768	2.37	-	22500
C1	31.5	7.	10.55	1.	920	.60	4500	4500		A0-15-3c	42.	6.	11.75	.375	14477	2.37	21500	22000
C2	31.5	7.	10.70	1.	880	.62	5500	5500		A0-11-2	29.375	6.	11.75	.375	12498	2.37	20000	25000
C3	31.5	7.	10.75	1.	1000	.60	5700	5700										
C4	31.5	7.	10.80	1.	980	.62	5650	5650										
Taylor, R. and Breuer, R. S. (1963) (*)																		
A1	33.	7.35	8.69	.75	4060	1.24	10864	10864		B1	35.43	9.45	11.81	1.18	4054	1.43	-	15763
A2	33.	7.35	8.69	.75	4180	1.24	10304	10304		B2	70.87	9.45	23.62	1.18	5177	2.86	-	26345
A3	33.	7.40	8.69	.75	4060	1.24	10416	10416		B3	106.30	9.45	35.43	1.18	4807	4.29	-	36376
A4	33.	7.35	8.69	.75	4260	1.24	8960	8960		B4	141.70	9.45	47.24	1.18	4409	5.72	-	39683
A5	33.	7.38	8.69	.75	5430	1.24	11984	11984		B5	70.87	9.45	23.62	1.18	4651	1.43	-	23369
A6	33.	7.38	8.69	.75	4520	1.24	11088	11088		B6	106.30	9.45	36.02	1.18	4324	1.35	-	25132
A7	33.	7.38	8.69	.75	4520	1.24	11088	11088		B7	106.30	9.45	4765	2.15	-	30313		
A8	33.	7.38	8.69	.75	5490	1.24	9744	9744		B8	106.30	9.45	36.02	1.18	4850	2.02	-	27557
A9	33.	7.40	8.75	.75	4260	.8	10416	10416										
A10	33.	7.40	8.75	.75	4180	.8	9072	9072		T2Ma	36.	6.	10.58	1.	4320	.88	-	8700
A11	33.	7.40	8.75	.75	5430	.8	11200	11200		T2Mb	36.	6.	10.58	1.	4020	.88	-	9225
A12	33.	7.40	8.75	.75	5490	.8	10304	10304		T2Mc	36.	6.	10.51	1.	4470	.88	-	11950
Bresler, Boris and Scordelis, A. C. (1963)																		
OA-1	72.0	12.2	18.15	.75	3270	4.	30000	37500										
OA-2	90.0	12.0	18.35	.75	3440	5.	32500	40000		R1	33.6	6.	10.	.75	3800	2.46	-	10100
OA-3	126.0	12.1	18.17	.75	5450	6.	35000	42500		R3	33.6	6.	10.	.75	3600	2.46	-	10100
Mattock, Alan H. (1969)																		
1	27.4	6.	10.	.75	2480	.62	8000	8200		D2	33.6	6.	10.	.75	4400	2.46	-	11800
3	27.4	6.	10.	.75	6800	.62	11500	12300										
10	27.4	6.	10.	.75	2690	1.86	11500	12600										
15	51.4	6.	10.	.75	3750	.62	7000	7000										
18	51.4	6.	10.	.75	2620	1.24	8000	8000										
22	51.4	6.	10.	.75	2340	1.86	9000	9000										
24	51.4	6.	10.	.75	4230	1.86	11800	11800										
Kreifeld, William J. and Thurston, Charles W. (1966)																		
4A3	36.	8.	15.36	1.	4440	2.54	22500	24700		D-1	49.5	9.	14.14	.75	8500	5.50	33000	34000
5A3	36.	8.	15.36	1.	4330	3.81	22500	38300		D-2	49.5	9.	14.14	.75	7330	5.50	29500	29500
11A2	36.	6.	12.36	1.	4380	2.54	15000	16500		D-3	49.5	9.	14.14	.75	6160	5.50	29000	29000
12A2	36.	6.	9.36	1.	4360	2.54	12500	14400		D-4	49.5	9.	14.14	.75	6060	5.50	30000	32500
18A2	36.	6.	12.44	1.	2800	2.00	13000	14200		D-5	49.5	9.	14.14	.75	7330	5.50	27500	29500
18B2	36.	6.	12.44	1.	2880	2.00	13000	16200		D-6	49.5	9.	14.14	.75	7050	5.50	28500	31500
18C2	36.	6.	12.44	1.	3280	2.00	12000	16500		D-7	49.5	9.	14.14	.75	5500	5.50	31500	31500
18D2	36.	6.	12.44	1.	3200	2.00	12000	13500		D-8	49.5	9.	14.14	.75	4350	5.50	25500	26500
13A2	36.	6.	12.56	1.	2890	.60	8500	10900		D-9	49.5	9.	14.14	.75	2570	5.50	20000	20000
14A2	36.	6.	9.56	1.	3000	.60	6000	7900		D-10	49.5	9.	14.14	.75	4550	5.50	28500	28500
15A2	36.	6.	12.44	1.	2920	1.00	9000	10300		D-11	49.5	9.	14.14	.75	3260	5.50	24500	24500
15B2	36.	6.	12.44	1.	3000	1.00	11000	11700		D-12	49.5	9.	14.14	.75	3970	5.50	22500	24000
16A2	36.	6.	9.44	1.	3220	1.00	8500	9400		D-13	49.5	9.	14.14	.75	3550	5.50	22300	22300
17A2	36.	6.	9.56	1.	3190	1.20	9000	9900		D-14	49.5	9.	14.14	.75	4080	5.50	24000	24000
18E2	36.	6.	12.44	1.	2870	2.00	12000	18400		D-15	49.5	9.	14.14	.75	3810	5.50	23000	23000
19A2	36.	6.	9.44	1.	2980	2.00	9500	16400		D-16	49.5	9.	14.14	.75	4420	5.50	25000	25000
20A2	36.	6.	9.36	1.	3050	2.54	10000	11400		D-17	49.5	9.	14.14	.75	3780	5.50	23500	23500
21A2	36.	8.	9.36	1.	2950	3.81	14000	17200		D-18	49.5	9.	14.14	.75	4160	5.50	23500	23500
2AC	48.	6.	10.00	1.	3340	.79	7000	8500		D-19	49.5	9.	14.14	.75	4680	5.50	26000	26000
3AC	48.	6.	10.06	1.	3020	1.20	9000	9900		D-20	49.5	9.	14.14	.75	4130	5.50	24000	24000
4AC	48.	6.	10.00	1.	2190	1.58	8500	8500		E-1	60.0	9.	14.14	.75	11220	5.50	33500	33500
5AC	48.	6.	9.94	1.	3210	2.54	12000	12000		E-2	60.0	9.	14.14	.75	8060	5.50	32500	32500
6AC	48.	6.	9.86	1.	3210	2.54	11500	11500		E-3	60.0	9.	14.14	.75	7060	5.50	29000	29000
7AC	48.	6.	10.00	1.	4235	1.58	12500	13000		E-4	60.0	9.	14.14	.75	3410	5.50	22000	22000
8AC	48.	6.	9.94	1.	4760	2.00	12000	12800		E-5	60.0	9.	14.14	.75	5790	5.50	40000	41500
9AC	48.	6.	10.00	1.	4420	1.58	11000	12100		A5-1	49.5	12.	14.14	.75	3470	4.68	22500	22500
10AC	48.	6.	9.94	1.	4760	2.00	11000	12200		A5-2	49.5	9.	14.39	.75	4010	3.44	21500	21500
11AC	48.	6.	9.86	1.	4950	2.54	12000	13300		A5-3	49.5	9.	14.52	.75	3150	5.50	24000	24000
12AC	48.	6.	10.00	1.	5570	1.58	11000	11800		A5-4	49.5	9.	14.47	.75	3900	3.17	21500	21500
13AC	48.	6.	9.94	1.	5430	2.00	9000	12000		A5-5	39.0	9.	14.56	.75	4260	2.26	23000	23000
14AC	48.	6.	9.86</															

Table 1 (cont.)—Test data from the literature

Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (in ²)	V ^c (lb)	V _c (lb)	Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (in ²)	V ^c (lb)	V _c (lb)		
2) Without stirrups, 4-point-bent																			
V-14	24.	8.	15.86	1.	3870	.954	20000	50350		267	37.38	6.04	10.60	.75	3000	.330	-	5500	
VI-15	24.	8.	15.86	1.	3700	.951	20000	40350		268	32.12	6.03	10.84	.75	2910	.321	-	6115	
VI-16	24.	8.	15.86	1.	3310	.951	20000	42400		269	10.62	6.05	10.77	.75	2620	.321	-	20000	
IIIe-17	60.	8.	15.86	1.	4240	3.23	17500	19800		270	21.35	6.00	10.74	.75	2910	.321	-	9300	
IIIe-18	60.	8.	15.86	1.	3650	3.23	16500	18150		246	37.49	6.03	10.80	.75	4000	.330	-	5700	
Ve-19	60.	8.	15.86	1.	3410	1.18	12000	14225		248	26.70	6.03	11.10	.75	4000	.330	-	8350	
Ve-20	60.	8.	15.86	1.	3710	1.18	12500	14825		249	10.62	6.04	10.85	.75	4060	.321	-	18800	
VIB-21	45.	8.	15.86	1.	3790	1.07	12500	16050		250	16.00	6.00	10.77	.75	4060	.321	-	12300	
VIB-22	45.	8.	15.86	1.	3740	1.07	12000	14025		251	21.40	6.06	10.86	.75	3800	.319	-	9420	
VIB-23	45.	8.	15.86	1.	4630	1.07	13500	16875		174	10.62	6.01	10.70	.75	5280	.330	-	19500	
VIB-24	60.	8.	15.86	1.	3820	.594	10500	12250		178	16.00	6.04	10.60	.75	5000	.330	-	11250	
VIB-25	60.	8.	15.86	1.	3740	.594	9000	11225		179	26.70	6.03	10.40	.75	4690	.330	-	7550	
Taylor, R. (1960)										180	37.38	6.04	10.60	.75	5000	.330	-	5600	
Ahmed, Shuaib H., Khaloo, A. R. and Proveda, A. (1986) (*)										181	21.40	5.94	10.63	.75	2800	.511	-	10940	
A1	32.	5.	8.	.5	9590	1.58	13000	13000		182	32.10	5.88	10.63	.75	2850	.491	-	7375	
A2	24.	5.	8.	.5	9590	1.58	14000	15500		183	21.36	6.05	10.80	.75	2560	.486	-	6795	
A3	21.6	5.	8.	.5	9590	1.58	14000	15500		184	32.04	6.11	10.78	.75	4270	.486	-	8720	
A4	18.4	5.	8.	.5	9590	1.58	14300	21000		185	42.72	6.05	10.60	.75	3670	.486	-	7560	
A5	16.	5.	8.	.5	9590	1.58	17000	37500		186	26.75	6.00	10.70	.75	3800	.496	-	9335	
A6	8.	5.	8.	.5	9590	1.58	34500	90000		187	26.70	6.05	10.54	.75	4170	.486	-	10030	
A7	32.76	5.	8.19	.5	9590	.71	8500	10500		188	53.40	6.08	10.51	.75	3850	.486	-	5765	
A8	24.57	5.	8.19	.5	9590	.71	9500	11000		189	16.02	6.03	10.67	.75	3630	.486	-	16150	
A9	22.11	5.	8.19	.5	9590	.71	11000	18000		190	26.70	6.06	10.71	.75	3920	.491	-	9735	
A10	18.84	5.	8.19	.5	9590	.71	11000	18500		191	26.70	6.02	10.75	.75	3920	.491	-	8850	
A11	16.38	5.	8.19	.5	9590	.71	12000	12500		192	16.05	6.00	10.78	.75	3700	.500	-	19600	
A12	8.19	5.	8.19	.5	9590	.71	30000	50000		193	21.04	6.03	10.62	.75	3700	.512	-	13800	
B1	31.76	5.	7.94	.5	10560	2.	11505	11510		194	26.75	6.00	10.69	.75	3800	.494	-	10185	
B2	23.82	5.	7.94	.5	10560	2.	12750	15500		195	32.10	6.00	10.67	.75	3830	.497	-	8830	
B3	21.44	5.	7.94	.5	10560	2.	14000	22500		196	21.36	6.04	10.71	.75	4900	.498	-	13250	
B4	18.26	5.	7.94	.5	10560	2.	14000	32200		197	21.36	6.06	10.51	.75	4900	.487	-	13950	
B5	15.88	5.	7.94	.5	10560	2.	17500	24000		198	26.70	6.14	10.73	.75	5130	.498	-	9100	
B7	32.76	5.	8.19	.5	10560	.9	10000	10030		199	26.70	5.97	10.70	.75	5130	.498	-	8550	
B8	24.57	5.	8.19	.5	10560	.9	10500	10500		200	42.72	6.15	10.67	.75	4900	.476	-	8050	
B9	22.11	5.	8.19	.5	10560	.9	10507	18000		201	32.10	5.98	10.68	.75	5130	.499	-	9050	
B10	18.84	5.	8.19	.5	10560	.9	12500	14400		202	32.10	6.05	10.80	.75	5130	.496	-	8600	
B11	16.38	5.	8.19	.5	10560	.9	14000	27500		203	32.10	6.00	10.72	.75	2950	.1.19	-	11000	
B12	8.19	5.	8.19	.5	10560	.9	25000	48000		204	42.80	5.99	10.85	.75	2880	.1.18	-	8725	
C1	29.00	5.	7.25	.5	10140	2.4	12000	12200		205	42.72	6.12	10.68	.75	2230	.1.17	-	8500	
C2	21.75	5.	7.25	.5	10140	2.4	11000	17000		206	53.40	6.08	10.68	.75	2230	.1.17	-	7200	
C3	19.58	5.	7.25	.5	10140	2.4	9000	15500		207	32.04	6.12	10.72	.75	2360	.1.17	-	9600	
C4	16.68	5.	7.25	.5	10140	2.4	12500	20000		208	16.02	6.09	10.81	.75	2550	.1.17	-	32200	
C7	32.52	5.	8.13	.5	10140	1.32	8000	10200		209	58.74	6.04	10.85	.75	2610	.1.17	-	9000	
C8	24.39	5.	8.13	.5	10140	1.32	10000	10000		210	26.75	5.95	10.79	.75	2630	.1.19	-	11140	
C9	21.95	5.	8.13	.5	10140	1.32	10000	10200		211	26.75	6.05	10.66	.75	2630	.1.17	-	11680	
C10	18.70	5.	8.13	.5	10140	1.32	9250	12800		212	53.50	6.06	10.76	.75	2880	.1.18	-	8660	
C11	16.26	5.	8.13	.5	10140	1.32	14500	24000		213	21.40	6.06	10.75	.75	2530	.1.18	-	13450	
C12	8.13	5.	8.13	.5	10140	1.32	20000	55000		214	21.40	5.88	10.79	.75	2520	.1.18	-	17250	
Kani, G. N. J. (1967)										215	21.36	6.00	10.68	.75	4040	.1.20	-	40900	
40	29.42	5.97	5.50	.75	3830	.05	-	7195		216	21.36	6.00	10.68	.75	3930	.1.20	-	17550	
41	13.38	6.00	5.56	.75	3950	.07	-	11565		217	26.70	6.00	10.68	.75	4320	.1.20	-	11550	
43	32.00	5.98	5.40	.75	4060	.08	-	6790		218	26.70	6.00	10.68	.75	4220	.1.20	-	12200	
45	10.70	5.95	5.23	.75	3700	.08	-	14520		219	48.06	6.00	10.68	.75	3560	.1.20	-	9650	
46	10.70	5.95	5.35	.75	3700	.08	-	15520		220	37.50	6.11	10.61	.75	3780	.1.18	-	10100	
47	26.70	5.95	5.20	.75	3590	.08	-	6335		221	32.04	6.06	10.83	.75	4930	.1.18	-	11600	
48	26.70	5.95	5.25	.75	3590	.08	-	6095		222	21.36	6.06	10.69	.75	3560	.1.20	-	23400	
52	21.40	6.00	5.45	.75	3600	.08	-	6495		223	21.36	6.00	10.68	.75	3930	.1.20	-	14650	
53	5.34	5.95	5.20	.75	3870	.08	-	34900		224	53.40	6.09	10.57	.75	4920	.1.16	-	10960	
54	5.34	5.95	5.35	.75	3870	.08	-	35450		225	16.02	6.05	10.67	.75	5090	.1.16	-	36700	
55	16.00	5.92	5.30	.75	3640	.08	-	7325		226	42.72	6.09	10.70	.75	5090	.1.16	-	12450	
56	18.73	6.03	5.41	.75	3950	.07	-	6295		227	21.36	6.04	10.90	.75	4800	.1.16	-	20800	
57	29.42	6.03	5.46	.75	3830	.06	-	7005		228	32.10	6.05	10.83	.75	4930	.1.18	-	11940	
58	18.73	6.00	5.45	.75	3950	.06	-	6500		229	26.70	6.04	10.93	.75	5020	.1.19	-	12750	
59	14.68	6.08	5.50	.75	3860	.08	-	11275		230	32.04	6.06	10.93	.75	5020	.1.19	-	11500	
60	16.02	6.10	5.46	.75	3880	.08	-	8835		231	42.72	6.03	10.84	.75	5020	.1.19	-	10625	
81	64.08	6.04	10.80	.75	3990	1.80	-	11500		232	26.75	5.92	10.77	.75	5220	.1.17	-	12000	
83	32.04	6.14	10.68	.75	3980	1.80	-	14600		233	79.77	6.00	10.75	.75	5220	.1.18	-	17250	
84	42.72	5.95	10.67	.75	3980	1.80	-	12450		234	17.72	3.94	5.91	.75	6699	.925	-	14550	
91	64.08	6.00	10.58	.75	3980	1.74	-	12095		235	35.43	5.91	11.81	.75	6699	.925	-	22817	
93	21.36	6.03	21.27	.75	3970	3.48	-	22950		236	53.15	7.87	17.72	.75	6699	1.851	-	34171	
94	21.36	6.03	21.37	.75	3970	3.48	-	131600		237	70.87	8.86	23.87	.75	6699	2.776	-		
71	64.08	6.10	21.42	.75	3970	3.48	-	22950		238	S-1	40.	6.06	10.18	.5	5300	.88	-	8000
72	42.80	6.00	21.62	.75	3600	3.51	-	44250		239	S-2	40.	6.06	10.44	.5	4800	.62	-	8400
74	64.20	6.00	20.60	.75	3950	3.51	-	24200		240	S-3	44.	6.00	10.50	.5	4200	.51	-	6990
75	64.20	6.00	20.63	.75	3960	3.51	-	24250		241	S-4	44.	6.00	10.56	.5				

Table 1 (cont.)—Test data from the literature

Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (in ²)	V ^c (lb)	V _c (lb)	Beam No.	a (in)	b (in)	d (in)	d _s (in)	f _c (psi)	A _s (%)	P _c (lb)	V _c (lb)	f _{yy} (psi)								
2) Without stirrups, 4-point-bent																										
C5	47.56	3.94	9.15	.375	4060	.48	-	6060	29a-1	72.	10.	17.94	1.	5630	4.	.111	35900	49500								
C6	27.56	3.94	9.15	.10	5220	.487	-	6180	29a-2	72.	10.	17.94	1.	5460	4.	.111	36000	49500								
D1	16.54	2.36	5.49	.10	5800	.122	-	2620	213.5-1	72.	10.	17.94	1.	5640	4.	.074	33300	49500								
D2	16.54	2.36	5.49	.10	5800	.122	-	2720	29a-2	72.	10.	17.94	1.	5390	4.	.111	48700	54000								
D3	16.54	2.36	5.49	.10	5800	.122	-	2380	29b-2	72.	10.	17.94	1.	6000	4.	.111	45500	54000								
D4	16.54	2.36	5.49	.10	5800	.122	-	2560	29c-2	72.	10.	17.94	1.	3500	4.	.111	36300	54000								
Chana, P. S. (1981)									29d-2	72.	10.	17.94	1.	4410	4.	.111	37100	54000								
2.1a	42.05	8.	14.02	.787	7150	1.89	-	21582	29e-2	72.	10.	17.94	1.	7030	4.	.111	46400	54000								
2.1b	42.05	8.	14.02	.787	7150	1.89	-	21829	213.5a-2	72.	10.	17.94	1.	5360	4.	.074	36300	54000								
2.2a	42.05	8.	14.02	.394	6034	1.89	-	19648	218a-2	72.	10.	17.94	1.	5450	4.	.095	36900	54000								
2.2b	42.05	8.	14.02	.394	6034	1.89	-	21222	29-3	72.	10.	17.94	1.	4970	4.	.111	40000	54000								
2.3a	42.05	8.	14.02	.787	6556	1.89	-	22346	318-1	72.	10.	17.94	1.	5880	4.	.122	49500	54000								
2.3b	42.05	8.	14.02	.787	6556	1.89	-	21672	321-1	72.	10.	17.94	1.	5620	4.	.105	36800	54000								
3.1a	20.91	3.94	6.97	.394	5004	.479	-	5350	318-2	72.	10.	17.94	1.	5640	4.	.122	39800	54000								
3.1b	20.91	3.94	6.97	.394	5004	.479	-	5373	321-2	72.	10.	17.94	1.	5510	4.	.105	37500	54000								
3.2a	20.91	3.94	6.97	.394	5337	.479	-	5508	313.5-3	72.	10.	17.94	1.	6190	4.	.163	48000	54000								
3.2b	20.91	3.94	6.97	.394	5337	.479	-	5733	318-3	72.	10.	17.94	1.	6240	4.	.122	39300	54000								
3.3a	20.91	3.94	6.97	.394	5816	.479	-	5957	321-3	72.	10.	17.94	1.	6240	4.	.105	31600	54000								
3.3b	20.91	3.94	6.97	.394	5816	.479	-	5216	Places, Alexander and Regan, Paul E. (1971)																	
D1	20.91	3.94	6.97	.394	4583	.479	-	4968	R8	33.6	6.	10.	.75	3870	.876	.21	17900	39150								
D2	20.91	3.94	6.97	.394	4699	.479	-	5261	R9	33.6	6.	10.	.75	4290	.876	.41	23500	40600								
D3	20.91	3.94	6.97	.394	6483	.479	-	4811	R10	33.6	6.	10.	.75	4295	.585	.21	16900	39150								
4.1a	12.52	2.36	4.17	.197	4482	.169	-	2208	R11	33.6	6.	10.	.75	3800	1.17	.21	20100	39150								
4.1b	12.52	2.36	4.17	.197	4482	.169	-	1945	R12	36.0	6.	10.	.75	4920	2.5	.21	24600	39150								
4.2a	12.52	2.36	4.17	.197	4482	.169	-	2017	R13	36.0	6.	10.	.75	4680	2.5	.41	33600	40600								
4.3a	12.52	2.36	4.17	.197	7571	.169	-	2630	R14	33.6	6.	10.	.75	4210	.876	.14	20100	39150								
4.3b	12.52	2.36	4.17	.197	7571	.169	-	2776	R15	36.0	6.	10.	.75	4330	2.5	.41	31400	40600								
4.4a	12.52	2.36	4.17	.197	7571	.169	-	2163	R16	36.0	6.	10.	.75	4580	2.5	.41	31400	40600								
5.1a	20.08	7.87	6.65	.394	5845	.946	-	10746	R17	33.6	6.	10.	.75	3850	.876	.21	15700	39150								
5.1b	20.08	7.87	6.65	.394	5845	.946	-	10746	R18	33.6	6.	10.	.75	4540	.876	.21	19000	39150								
5.2a	20.08	7.87	6.65	.394	5758	.946	-	12364	R19	33.6	6.	10.	.75	4390	.876	.41	26900	40600								
5.2b	20.08	7.87	6.65	.394	5758	.946	-	12589	R20	33.6	6.	10.	.75	6230	.876	.21	20200	39150								
6.1	4.96	.925	1.65	.094	5192	.026	-	481	R21	36.0	6.	10.	.75	6980	2.5	.41	32600	40600								
6.2	4.96	.925	1.65	.094	5511	.026	-	432	R22	45.0	6.	10.	.75	4280	.876	.21	17900	39150								
6.3	4.96	.925	1.65	.094	5671	.026	-	472	R24	50.5	6.	10.	.75	4480	2.5	.21	20700	39150								
6.4	4.96	.925	1.65	.094	8702	.026	-	470	R25	36.0	6.	10.	.75	4470	2.5	.21	23500	39150								
6.5	4.96	.925	1.65	.094	6802	.026	-	587	R27	36.0	6.	10.	.75	1980	2.5	.41	21300	40600								
6.6	4.96	.925	1.65	.094	9282	.026	-	516	R28	36.0	6.	10.	.75	4580	2.5	.83	40300	39150								
6.7	4.96	.925	1.65	.094	6251	.026	-	531	Mattock, Alan H. and Wang, Zuhua (1984) (*)																	
6.8	4.96	.925	1.65	.094	6179	.026	-	504																		
6.9	4.96	.925	1.65	.094	6527	.026	-	515																		
Walraven, J. C. (1978) (*)																										
A1	14.76	7.87	4.92	.63	4978	.321	-	6704	Bresler, Boris and Scordella, A. C. (1963)																	
A2	49.61	7.87	16.54	.63	4959	.964	-	15882	A-1	72.	12.1	18.35	.75	3490	4.	.1	52500	47200								
A3	85.04	7.87	28.35	.63	5041	1.77	-	22576	A-2	90.	12.	18.27	.75	3520	5.	.1	55000	47200								
A4	Elzanty, Ashraf H., Nilsson, Arthur H., and slate, F. O. (1986)																									
F7	42.0	7.	10.7	.5	3000	.45	-	7600	A-3	126.	12.1	18.35	.75	5080	6.	.1	52500	47200								
F11	42.8	7.	10.7	.5	3000	.90	-	10000	B-1	72.	9.1	18.15	.75	3590	4.	.147	50000	47200								
F12	42.0	7.	10.5	.5	3000	1.80	-	12000	B-2	90.	9.	18.33	.75	3360	4.	.148	45000	47200								
F8	42.8	7.	10.7	.5	5800	.75	-	10300	B-3	126.	9.	18.13	.75	5620	5.	.148	40000	47200								
F13	42.8	7.	10.7	.5	5800	.90	-	10400	C-1	72.	6.1	18.25	.75	4290	2.	.199	35000	47200								
F14	42.0	7.	10.5	.5	5800	1.80	-	14300	C-2	90.	6.	18.28	.75	3450	4.	.202	36500	47200								
F1	42.8	7.	10.7	.5	9500	2.40	-	17300	C-3	126.	6.1	18.06	.75	5080	4.	.199	30500	47200								
F9	42.4	7.	10.6	.5	11500	1.20	-	14100	Clark, Arthur F. (1951)																	
F15	42.0	7.	10.5	.5	11500	1.80	-	15000																		
F3	21.4	7.	10.7	.5	10000	.90	-	18100	E1-1	36.	8.	15.36	.38	3430	3.81	.38	47016	48020								
F4	21.0	7.	10.5	.5	10000	1.80	-	25600	E1-2	36.	8.	15.36	.38	3395	3.81	.38	50016	48020								
F5	64.42	7.	10.2	.5	9200	.90	-	9800	E1-3	36.	8.	15.36	.38	3590	3.81	.38	55016	48020								
F6	63.0	7.	10.5	.5	9200	1.80	-	13600	5) With stirrups, 4-point-bent																	
Xuech, H. et al. (1962) (*) (*)																		Johnston, Bruce and Cox, Kenneth C. (1939)								
X	15.75	3.54	5.40	.75	3421	.41	-	3285	B-1-I	36.	12.	12.	.75	3190	1.12	.104	22600	45000								
Y	28.22	4.72	7.84	.59	3421	.982	-	6768	B-1-II	36.	12.	12.	.75	319												

Table 1 (cont.)—Test data from the literature

Beam No.	a (in)	b (in)	d (in)	d_s (in)	f_c (psi)	A_g (in 2)	P_v (%)	V_u (lb)	f_{vv} (psi)	Beam No.	a (in)	b (in)	d (in)	d_s (in)	f_c (psi)	A_g (in 2)	P_v (%)	V_u (lb)	f_{vv} (psi)								
5) With stirrups, 4-point-bent																											
2-I 3/8	32.	5.5	18.25	.75	4185	4.	.743	84000	47900	B50	61.	7.5	15.39	.75	4390	575	.104	24050	73600								
1-D 3/8	32.	5.5	18.25	.75	2855	4.	.615	64875	47900	C00	61.	7.5	15.41	.75	4270	1.09	0.	13270	-								
2-D 3/8	32.	5.5	18.25	.75	3605	4.	.615	71825	47900	C25	61.	7.5	15.33	.75	4100	1.09	.052	18650	62100								
1a-v 1/4	32.	5.5	19.5	.75	3540	2.	.28	53050	46000	C50	61.	7.5	15.47	.75	4300	1.09	.104	30150	73600								
1a-u 3/8	32.	5.5	19.5	.75	3335	2.	.615	58675	52000	C75	61.	7.5	15.57	.75	4260	1.09	.157	31020	65700								
Clark, Author P. (1951)										Placitas, Alexander and Regan, Paul E. (1971) (3-point Bent)																	
B1-1	30.	8.	15.36		3388	3.81	.37	62695	48020	T1	33.6	6.	10.	.75	4050	.75	.21	24700	39150								
B1-2	30.	8.	15.36		3680	3.81	.37	57685	48020	T2	33.6	6.	10.	.75	4070	.876	0.	12300	-								
B1-3	30.	8.	15.36		3435	3.81	.37	64030	48020	T3	33.6	6.	10.	.75	3990	.876	.21	23500	39150								
B1-4	30.	8.	15.36		3380	3.81	.37	60280	48020	T4	33.6	6.	10.	.75	4710	1.17	.21	24600	39150								
B1-5	30.	8.	15.36		3570	3.81	.37	54280	48020	T5	33.6	6.	10.	.75	4890	.876	.41	31400	40600								
B2-1	30.	8.	15.36		3370	3.81	.73	67685	48020	T6	36.0	6.	10.	.75	3470	2.5	.83	46000	39150								
B2-2	30.	8.	15.36		3820	3.81	.73	72435	48020	T7	34.6	6.	10.4	.75	3970	1.8	.21	24600	39150								
B2-3	30.	8.	15.36		3615	3.81	.73	75280	48020	T8	36.0	6.	10.	.75	4530	2.5	.21	28000	39150								
B6-1	30.	8.	15.36		6110	3.81	.37	65280	48020	T9	36.0	6.	10.	.75	2930	2.5	.41	34700	40600								
C1-1	24.	8.	15.36		3720	2.54	.34	62435	48020	T10	33.6	6.	10.	.75	4090	.876	.14	19500	39150								
C1-2	24.	8.	15.36		3820	2.54	.34	69935	48020	T11	36.0	6.	10.	.75	5360	2.5	.41	36000	40600								
C1-3	24.	8.	15.36		3475	2.54	.34	55280	48020	T12	36.0	6.	10.	.75	4450	2.5	.21	32500	39150								
C1-4	24.	8.	15.36		4210	2.54	.34	64280	48020	T13	33.6	6.	10.	.75	1850	.876	.21	20200	39150								
C2-1	24.	8.	15.36		3430	2.54	.69	65185	48020	T14	36.0	6.	10.	.75	4840	2.5	.83	49300	39150								
C2-2	24.	8.	15.36		3625	2.54	.69	67685	48020	T15	72.0	6.	10.	.75	4810	2.5	.21	23500	39150								
C2-4	24.	8.	15.36		3910	2.54	.69	64780	48020	T16	72.0	6.	10.	.75	4740	2.5	.14	20800	39150								
C3-1	24.	8.	15.36		2040	2.54	.34	50280	48020	T17	72.0	6.	10.	.75	4790	2.5	.41	30100	40600								
C3-2	24.	8.	15.36		2000	2.54	.34	45030	48020	T18	33.6	6.	10.	.75	4980	.876	0.	16800	-								
C3-3	24.	8.	15.36		2020	2.54	.34	42200	48020	T19	54.0	6.	10.	.75	4340	2.5	.21	25500	39150								
C4-1	24.	8.	15.36		3550	3.81	.34	69530	48020	T20	54.0	6.	10.	.75	4655	2.5	.41	34600	40600								
C6-2	24.	8.	15.36		6560	3.81	.34	95280	48020	T22	33.6	6.	10.	.75	4980	.876	.21	24600	39150								
C6-3	24.	8.	15.36		6480	3.81	.34	97780	48020	T25	33.6	6.	10.	.75	7840	.876	.21	25800	39150								
C6-4	24.	8.	15.36		6900	3.81	.34	96359	48020	T26	36.0	6.	10.	.75	8260	2.5	.41	40300	40600								
D1-1	18.	8.	15.37		3800	2.	.46	67685	48020	T27	36.0	6.	10.	.75	1740	2.5	.41	29700	40600								
D1-3	18.	8.	15.37		3560	2.	.46	57685	48020	T31	33.6	6.	10.	.75	4495	.876	.21	21300	39150								
D2-1	18.	8.	15.37		3480	2.	.61	65185	48020	T32	36.0	6.	10.	.75	4000	2.5	.83	48600	39150								
D2-2	18.	8.	15.37		3755	2.	.61	70185	48020	T34	54.0	6.	10.	.75	4920	2.5	.21	25200	39150								
D3-1	18.	8.	15.37		4090	3.	.92	80781	48020	T35	54.	6.	10.	.75	4880	2.5	.21	25800	39150								
D4-1	18.	8.	15.37		3350	2.	1.22	70185	48020	T36	36.	6.	10.	.75	3500	2.5	.41	40300	40600								
D1-6	24.	6.	12.36		4010	2.54	.46	39281	48020	T37	36.	6.	10.	.75	4615	2.5	.83	47100	39150								
D1-7	24.	6.	12.36		4060	2.54	.46	40281	48020	T38	36.	6.	10.	.75	4380	2.5	.83	53800	39150								
D1-8	24.	6.	12.36		4030	2.54	.46	41781	48020	Haddadin, Munther J., Hong, Sheu-Tien and Mettlock, Alan H. (1971)																	
E1-2	25.	6.	12.36		4375	2.54	.73	98959	48020	A1	37.5	7.	15.		4275	4.	0.	26250	-								
D2-6	30.	6.	12.36		4280	2.54	.61	37859	48020	A2	37.5	7.	15.		4235	4.	.19	43680	52000								
D2-7	30.	6.	12.36		4120	2.54	.61	35359	48020	A3	37.5	7.	15.		4360	4.	.419	65520	50000								
D2-8	30.	6.	12.36		3790	2.54	.61	37859	48020	A4	37.5	7.	15.		4145	4.	.786	76965	50000								
D4-1	30.	6.	12.36		3970	2.54	.49	37859	48020	A5	37.5	7.	15.		3815	4.	.126	87045	50000								
D4-2	30.	6.	12.36		3720	2.54	.49	35359	48020	B3'	50.625	7.	15.		4015	4.	.419	61000	50000								
D4-3	30.	6.	12.36		3200	2.54	.49	37109	48020	C1	63.75	7.	15.		3760	4.	0.	19625	-								
D5-1	30.	6.	12.36		4020	2.54	.37	32859	48020	C2	63.75	7.	15.		4030	4.	.19	39895	52000								
D5-2	30.	6.	12.36		4210	2.54	.37	35359	48020	C3	63.75	7.	15.		3500	4.	.419	58485	50000								
D5-3	30.	6.	12.36		3930	2.54	.37	35359	48020	C4	63.75	7.	15.		3730	4.	.786	69825	50000								
(*) concrete cubic strength										C5	63.75	7.	15.		4415	4.	.126	80535	50000								
6) With Stirrups, T-section										D3	90.	7.	15.		4245	4.	.419	54180	50000								
Palakas, Michael N., Darwin, Davis (1980) (3-point Bent)										E1	37.5	7.	15.		2020	4.	0.	22365	-								
A00	61.	7.5	15.54	.75	4740	.765	0.	16244	-	E2	37.5	7.	15.		2200	4.	.19	38010	52000								
A25	61.	7.5	15.38	.75	4720	.765	.052	19275	61000	E3	37.5	7.	15.		1945	4.	.786	56490	50000								
A25a	61.	7.5	15.26	.75	4790	.765	.052	20722	61000	F3	37.5	7.	15.		2475	4.	.126	69090	50000								
A50	61.	7.5	15.42	.75	3810	.765	.104	25954	71400	G3	37.5	7.	15.		6515	4.	.419	74130	50000								
A50a	61.	7.5	15.49	.75	4060	.765	.104	24660	72500	G4	37.5	7.	15.		3800	4.	.419	74760	66200								
A75	61.	7.5	15.56	.75	4670	.765	.157	31966	61900	G5	37.5	7.	15.		3880	4.	.629	86205	66200								
B00	61.	7.5	15.70	.75	4640	.575	0.	16027	-	H1	37.5	7.	15.		3790	4.	.105	96180	66200								
B25	61.	7.5	15.52	.75	4740	.575	.052	16760	62100	H2	37.5	7.	15.		4240	2.	.419	63735	50000								
										I1	37.5	7.	15.		4090	6.	.419	71610	50000								

$$1 \text{ in.} = 25.4 \text{ mm}; 1 \text{ in.}^2 = 645.2 \text{ mm}^2; 1 \text{ lb} = 4.448 \text{ N}; 1 \text{ psi} = 0.006895 \text{ MPa.}$$

Note: The data of Clark (1951), Haddadin (1971), and T-section data of Rüsch (1972) are not used due to lack of information.

$$v_u = v_s + v_c \quad (6)$$

v_s represents the contribution of the yield forces in stirrups, and v_c represents the additional contribution due to concrete. For the contribution of stirrups, plastic limit analysis readily yields the expression

$$v_s = \frac{A_v f_{yy}}{hs} (\sin\alpha + \cos\alpha) \quad (7)$$

where

A_s = cross-sectional area of the stirrup

s = spacing of the stirrups

b = beam width

f_{yv} = yield strength of stirrups

α = angle between the stirrups and the longitudinal axis of the beam

Although it has been recognized that the presence of stirrups has some beneficial effect on the magnitude of v_c , this effect is neglected in the current code specifications, i.e., v_c is taken to be the same as in a beam with no stirrups. The assumption that the magnitude of v_c is unaffected by the presence of stirrups is more reasonable when the design formula is based on the initial cracking. However, design on the basis of initial cracking appears to be insufficient because the ratio of the ultimate load to the initial cracking load decreases as the beam size increases, thus causing the safety margin against failure to be nonuniform. The ideal approach would be to use design criteria for both the ultimate load and the first cracking load. However, if only one of these two criteria should be used, then the failure load criterion is probably more reasonable from the safety viewpoint.

The reason that the stirrups enhance the strength contribution of concrete is probably due to several ef-

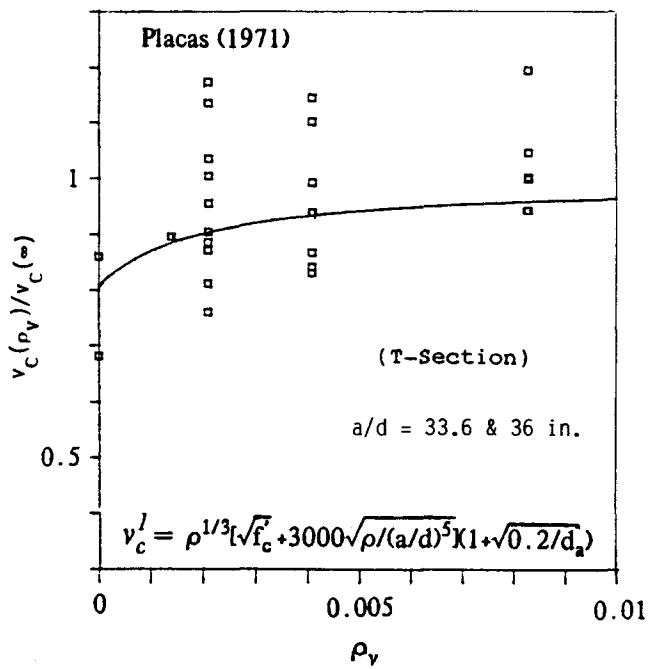


Fig. 4—Effect of stirrup reinforcement ratio on diagonal shear strength of beams without stirrups

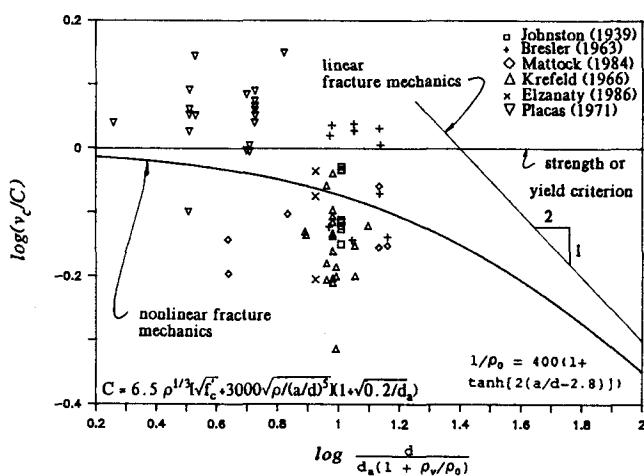


Fig. 5—Size effect in beams with stirrups and effect of stirrup ratio

fects. One of them is that cracks are forced to redistribute at a closer spacing and over a wider area. This results in a smaller crack width and therefore, in an increased shear and tension transfer capability. Another effect enhancing the beam strength arises from the fact that the stirrups support the longitudinal bar and thus prevent concrete splitting along the bar.

If one looks up the derivation of the size-effect law in Reference 42, it transpires that redistribution of cracking over a wider area should manifest itself in an increase of coefficient λ_0 in Eq. (1) or (3). Thus, from the viewpoint of the size effect, the proper way to model the influence of stirrups on the strength contribution of concrete appears to be an increase of coefficient λ_0 as a function of the stirrup-reinforcement ratio

ρ_v , as proposed by Palakas and Darwin.²⁹ For the sake of simplicity, we assume a linear dependence, realizing that the scatter of the test data makes it impossible to calibrate any more complicated dependence. So we may write

$$\lambda_0 = 25 \left(1 + \frac{\rho_v}{\rho_0} \right) \quad (8)$$

in which ρ_0 is an empirical coefficient indicating the stirrup reinforcement ratio ρ_v for which λ_0 is doubled compared to λ_0 for $\rho_v = 0$.

The increase of concrete shear capacity described by coefficient ρ can be realized only if the relative shear span is not too small; roughly $a/d \geq 1.5$. For smaller shear spans, the stirrups do not appear to develop their full yield capacity, as suggested by Zsutty.²¹ Therefore, the value of coefficient ρ_0 cannot be constant, but must change from 0 for $a/d < 1.5$ to a finite value for larger shear spans. The transition should, of course, be expected to be smooth rather than abrupt, and a smooth formula is also preferable for certain numerical procedures or computer optimization of design. A simple empirical formula that gives a smooth transition from a zero value to a finite value is

$$\frac{1}{\rho_0} = a_0 \left[1 + \tanh \left(2 \frac{a}{d} - 5.6 \right) \right] \quad (9)$$

in which a_0 is an empirical constant and coefficients 2 and 5.6 have been approximately determined by optimization of data fits.

The available experimental evidence on the effect of ρ_v on v_c is shown in Fig. 4, in which the ordinate is the ratio of the measured v_c to the value of v_c calculated for $\rho_v \rightarrow \infty$ (or $\lambda_0 \rightarrow \infty$). The curve represents the trend according to Eq. (8). Although the scatter is very large, the increasing trend is nevertheless clearly apparent. The effect of ρ_v is also evident from the regression in Fig. 5.

Fig. 5 shows the effect of beam depth d on the contribution of concrete to the shear capacity of a beam with stirrups. The existing test data, plotted in the figure, are those of Johnston and Cox,¹ Bresler and Scordelis,¹¹ Mattock and Wang,³³ Krefeld and Thurston,¹⁴ Elzanaty et al.,³⁶ and Placas and Regan.²⁴ The curve in Fig. 5 shows the present formula. Although the data scatter is very large, the presence of the size effect in beams with stirrups is nevertheless evident.

Fig. 6 shows, for beams with stirrups, a plot of the measured values of v_u versus the values calculated from the formulas. The plot in Fig. 6(c) is based on the presently proposed equations [Eq. (3), (2), (6), (8), (9), and (7)], and for comparison Fig. 6(d) shows the plot obtained when the effect of the stirrups is omitted, i.e., $1/\rho_0 = 0$. We see that by considering the effect of stirrups, the coefficient of variation of the deviations from the straight line is reduced from $\omega = 0.156$ to $\omega = 0.141$ for this set of 87 data points. The improvement due to considering the effect of stirrups on concrete ca-

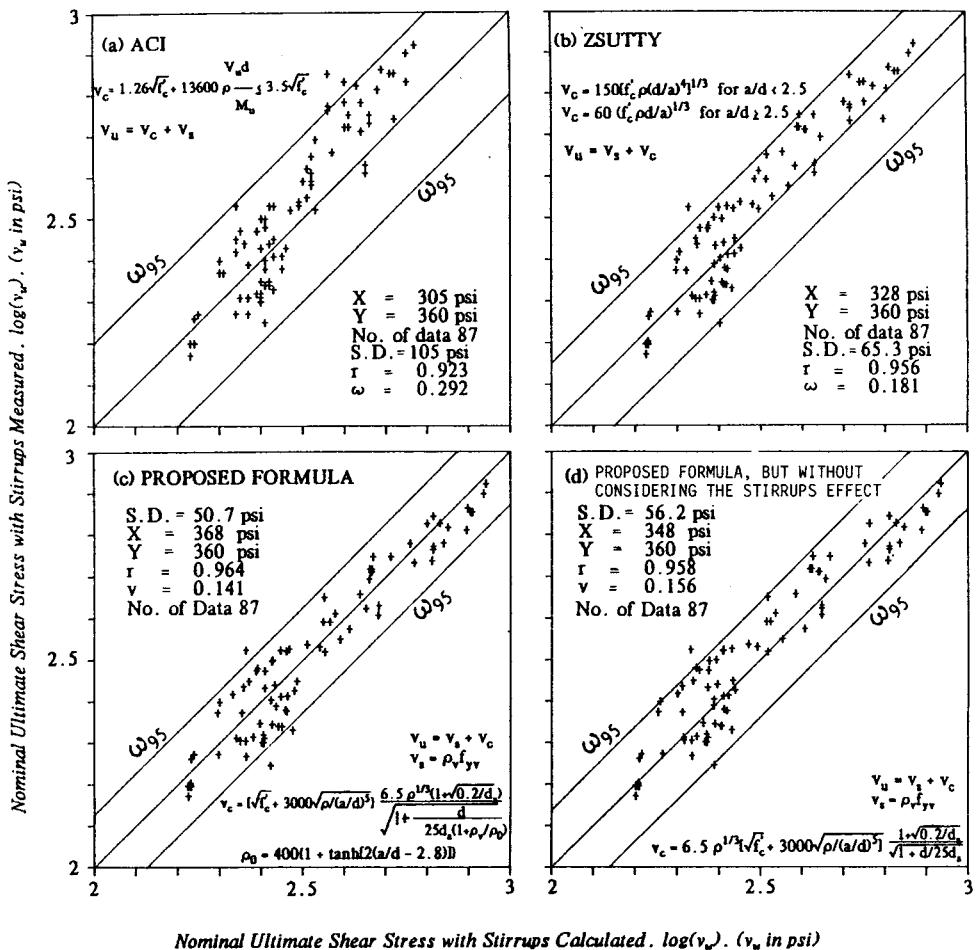


Fig. 6—Plots of measured versus calculated values of mean nominal shear strength for beams with stirrups (1 psi = 6895 Pa)

pacity is small but nevertheless appreciable. For comparison, Fig. 6(a) and (b) show the same plots for the current ACI specifications⁵¹ and for Zsutty's formula¹⁸ [also see Eq. (15) and (18) of Reference 36]. We see that the coefficients of variation of the vertical deviations from the straight line are for these two formulas 0.292 and 0.181, which is distinctly larger than the value for the presently proposed formula. Note also the increased separation of the 95 percent confidence limits.

The same set of 87 data points is plotted in Fig. 7 to show the size effect, i.e., the dependence of v_c on the relative size d/d_a . Comparison with the proposed formulas is shown in Fig. 7(c) and (d), and despite large scatter the presence of a downward trend representing a size effect is clearly visible. By contrast, the scatter in Fig. 7(a) for the ACI specification is so large that no size effect is easily detected. This means that it would make no sense to introduce the size effect into the ACI formula without improving the form of the existing formula itself, which was previously shown possible for beams without stirrups.⁴⁵

In all the preceding comparisons with tests, the formulas were made to represent the mean trend of the test data. The design formulas should, however, be introduced in such a manner that most of the test data are

on the safe side of the predicted values. This is achieved by multiplying the formula for the mean trend by a factor. According to the experimental evidence, a suitable adjustment is achieved by changing the value of k_1 in Eq. (2) from 6.5 to 4.5. The plot of the measured versus calculated values of the nominal shear strength v_u for beams with stirrups is shown (for the design formula with $k_1 = 4.5$) in Fig. 8. In these plots, the majority of the test data should lie above the inclined straight line, and we see that this is well satisfied for Fig. 8(c) and (d) for the present formulas. However, for the existing ACI formula there are more data points falling significantly below this line, and also more data points high above the straight line — a case that represents overdesign.

CONCLUSIONS

1. The previously proposed formula of Bažant and Kim⁴⁵ for diagonal shear failure of longitudinally reinforced beams without stirrups is improved by introducing, in addition to the effect of the relative beam size, the effect of the maximum aggregate size. The new formula is calibrated according to a larger set of test data than before, consisting of 461 test results that were compiled from the literature.

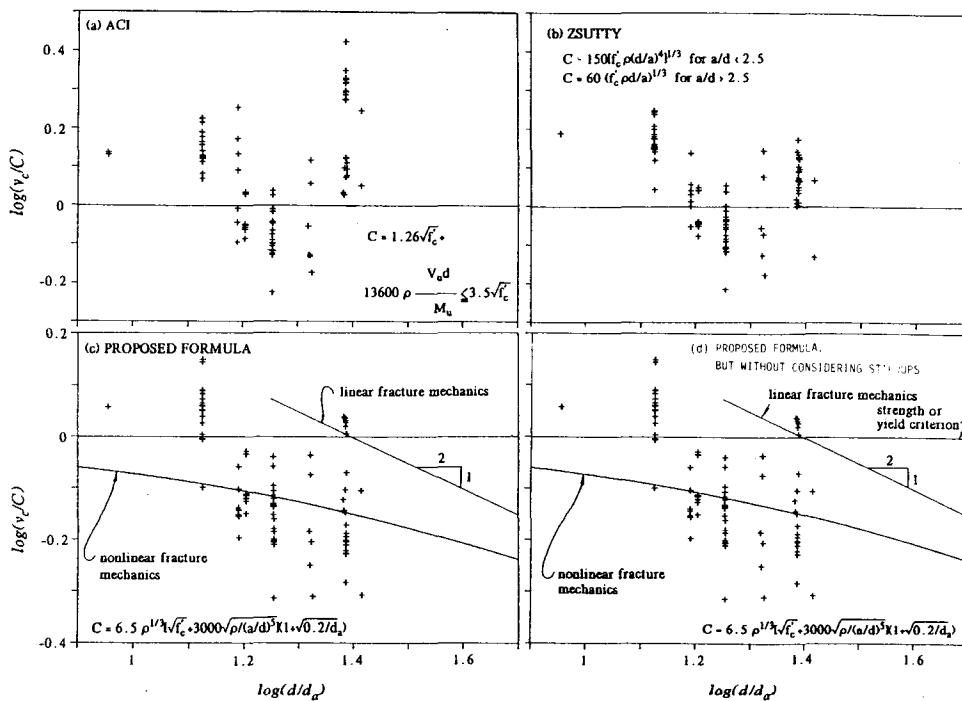


Fig. 7—Size-effect plots for various formulas for beams with stirrups in comparison with test data

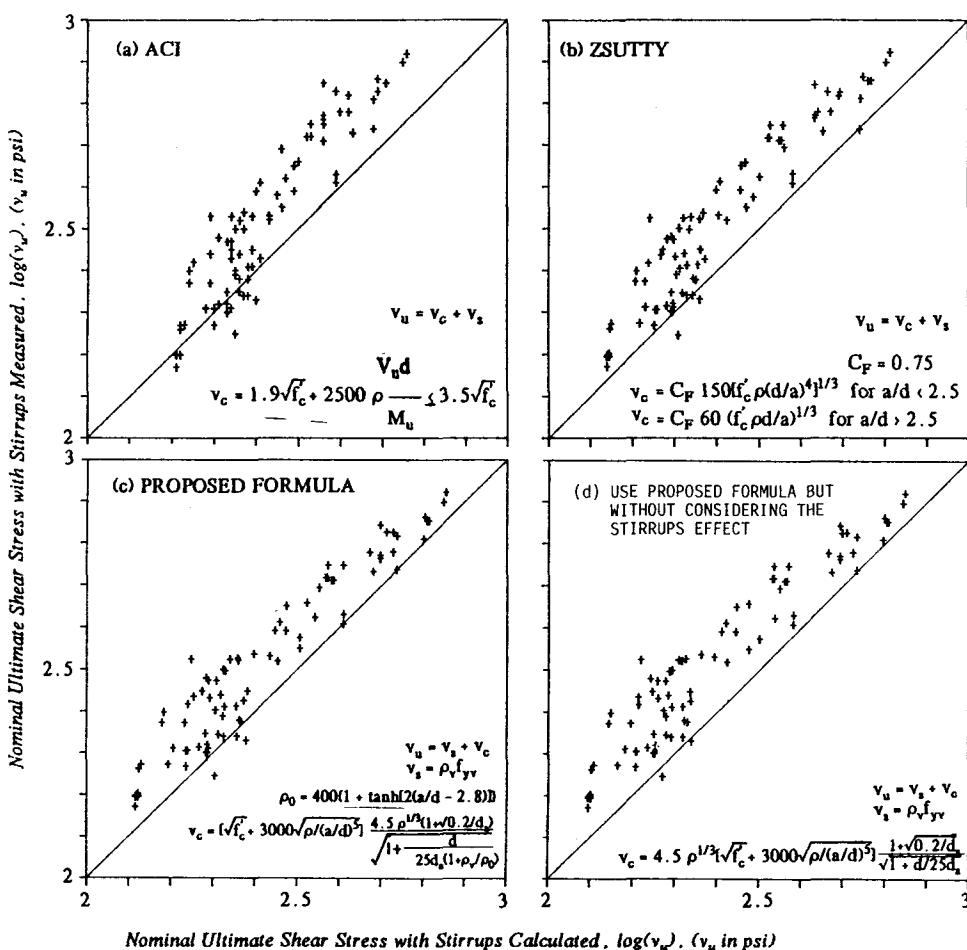


Fig. 8—Plots of measured versus calculated values of nominal shear strength for various design formulas

2. The formula is further extended to diagonal shear failure of reinforced concrete beams with stirrups. The generalization of the formula takes into account the fact that the presence of stirrups has a strengthening effect on the shear capacity of concrete. The degree of strengthening depends, however, on the shear span. The resulting formula is calibrated according to a set of 87 test results compiled from the literature. The results confirm that the size effect on the concrete shear strength still exists in the presence of stirrups, but it is milder than without stirrups.

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